

UNIVERSITY OF BOLTON

OFF CAMPUS DIVISION

WESTERN INTERNATIONAL COLLEGE FZE

BENG(HONS) CIVIL ENGINEERING

TRIMESTER TWO EXAMINATION 2021/2022

ADVANCED STRUCTURAL ANALYSIS AND DESIGN

MODULE NO. CIE6001

Date: Tuesday 26th April 2022

Time: 10:00am – 1:00pm

INSTRUCTIONS TO CANDIDATES:

There are **FIVE** questions on this paper.

Answer **ANY FOUR** questions.

All questions carry equal marks.

Marks for parts of questions are shown in brackets.

This examination paper carries a total of 100 marks.

Formula sheet / supplementary information is provided at the end of question paper.

All working must be shown. A numerical solution to a question obtained by programming an electronic calculator will not be accepted.

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Q1

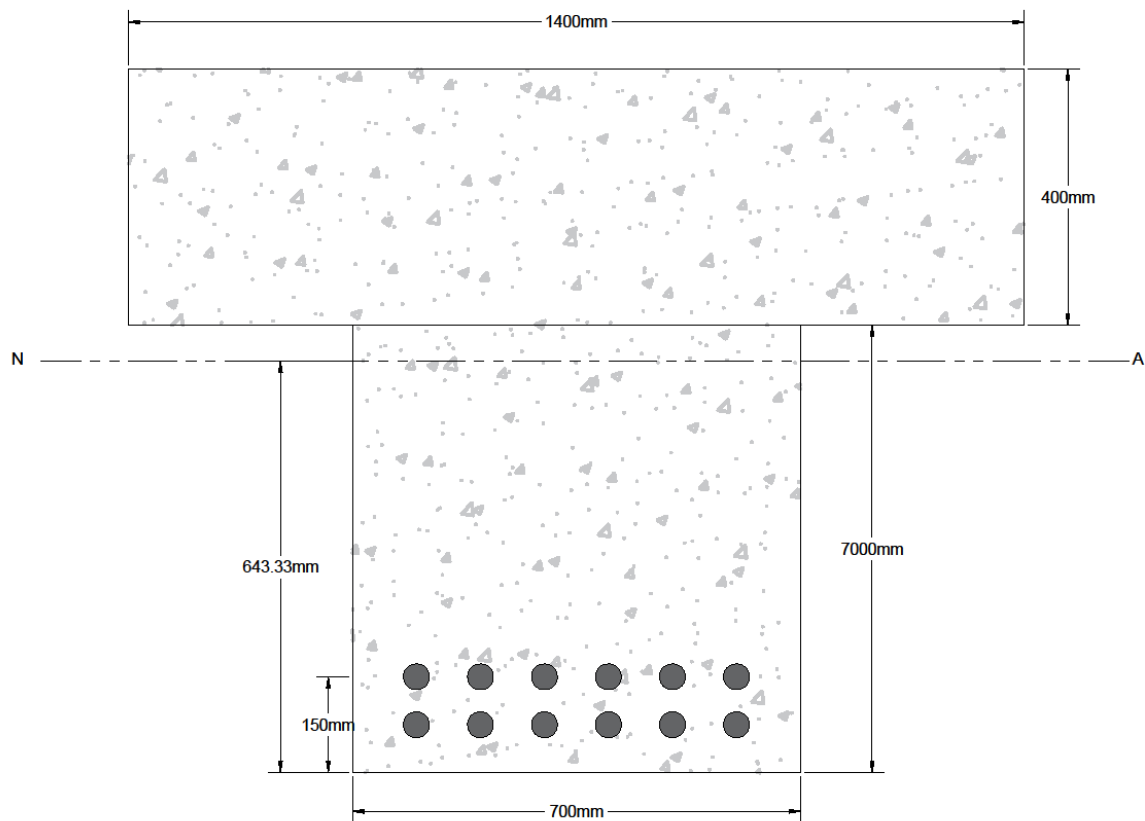


Figure Q1

Figure Q1 shows a T-shaped pre-stressed concrete beam section. The beam contains **twelve** pre-stressing strands (12.7mm diameter) at an average height of 150mm from the bottom of the beam.

The beam supports a parking area and so the proportion of the variable load to be considered in the quasi permanent loading condition is 0.6. In service, the beam is simply supported over a span of 8m and carries the following loads:

| | |
|---|---------|
| Permanent load (including beam self-weight) | 35 kN/m |
| Variable load | 50 kN/m |
| Characteristic breaking load of one strand | 184 kN |

Q1 continued over the page

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Q1 continued

| | |
|-------------------------------|------------------------------|
| Initial pre-stress | 70% of UTS |
| Pre-stress losses | 25% of initial pre-stress |
| Concrete strength at transfer | $f_{ck} = 40 \text{ N/mm}^2$ |
| Concrete strength in service | $f_{ck} = 55 \text{ N/mm}^2$ |

Limiting stresses in concrete:

| | | |
|-------------|-------------------------------|----------------------------------|
| At transfer | $0.6 f_{ck}$ in compression; | 1 N/mm^2 in tension |
| In service | $0.45 f_{ck}$ in compression; | 3.80 N/mm^2 in tension |

- a) Calculate the stresses in the concrete at the top and bottom of the beam:
- (i) At transfer; **(9 marks)**
 - (ii) In service under quasi-permanent loads **(6 marks)**
- b) Draw the distribution of stress over the height of the beam
- (i) At transfer; **(2.5 marks)**
 - (ii) In service under quasi-permanent loads **(2.5 marks)**
- c) Compare the calculated values of stress in the concrete with the limiting values of stress in the concrete:
- (i) at transfer; (ii) in service under quasi-permanent loads.

Comment on the adequacy of the beam.

(5 marks)

Total 25 marks

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Q2

- a) **Figure Q2(a)** shows the section of an internal steel column UKC 254x254x132 to be used in a multi story building. The column has pinned boundary conditions at each end; and the inter storey height is 5m.

By using the EC3 method, assess the suitability of the section to resist an ultimate design axial compressive load of 2800Kn.

(18 marks)

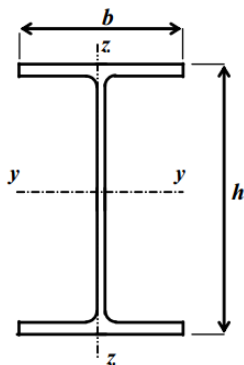


Figure Q2(a)

| | | |
|--|---|----------------------|
| h | = | 276.3mm |
| b | = | 261.3mm |
| tw | = | 15.3mm |
| tf | = | 25.3mm |
| A | = | 168cm ² |
| I _y | = | 22529cm ⁴ |
| I _z | = | 7531cm ⁴ |
| i _y | = | 11.6cm |
| i _z | = | 6.69cm |
| Class 1 section | | |
| Steel grade S275 | | |
| Modulus of Elasticity E = 210 kN/mm ² | | |
| Yield Strength fy = 275 N/mm ² | | |

Additional information:

Euler Critical load

$$N_{cr} = \frac{\pi^2 EI}{l_{cr}^2}$$

Design method and data sheet for buckling of columns to EC3 are attached at the end of this paper.

Q2 continued over the page

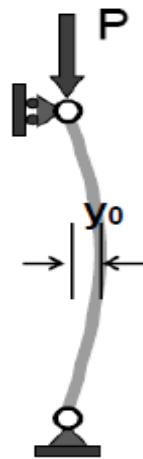
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Q 2 continued

b)

Figure Q2(b) shows a pin ended real strut made of steel strip 30mm x 4mm having a length equal to 150 mm. The strut has a small initial curvature causing a departure of $y_0 = 0.6$ mm at its mid length. If an axial load of 5.5 kN is applied to the strut compute the average stress in the strut.



| |
|---|
| $E = 200 \text{ kN/mm}^2$ $f_y = 275 \text{ N/mm}^2$ |
|---|

Figure Q2(b)

(7 marks)

Total 25 marks

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Q3

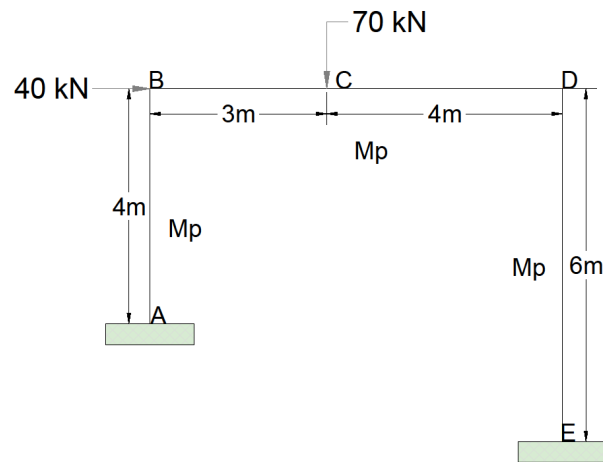


Figure Q3

Figure Q3 shows a rigid-jointed frame ABCDE consists of the beam BD of span 7m and columns of height 4m (Column AB) and 6m (Column DE). The columns are fixed at the base. The frame carries a vertical point load 70 kN, 3m away from B and a horizontal point load at B as depicted in **Figure Q3**.

a) Find the values of plastic moment which correspond to the following collapse mechanisms:

(i) Plastic hinges at B, C and D

(6 marks)

(ii) Plastic hinges at A, D and E

(6 marks)

(iii) Plastic hinges at A, B, D and E

(6 marks)

b) Draw the bending moment diagram for the most critical of the collapse mechanisms in part (a), showing values at A, B, C, D and E.

(7 marks)

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Q4

The L shaped bracket shown in **Figure Q4(a)** and **Q4(b)** is connected to a steel column 310mm deep with 8 Nos M20 grade 8.8 bolts. The bracket is formed from UB 409 x 178 x 74 kg/m steel section with the following properties:

| | |
|------------------|---------|
| Web thickness | 9.7mm |
| Flange thickness | 16mm |
| Depth of section | 412.9mm |
| Width of section | 179.7mm |

A factored vertical load 70 kN is applied at the location shown in the plan view of the bracket.

- (i) What is the out of plane moment in the bolt group?
(2.5 marks)
- (ii) What is the in plane moment in the bolt group?
(2.5 marks)
- (iii) What are the tension and the shear in the four bolts in bolt rows b1 and b3?
(15 marks)
- (iv) Comment on the adequacy of the specified bolts.
(5 marks)

Total 25 marks

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Q4 continued

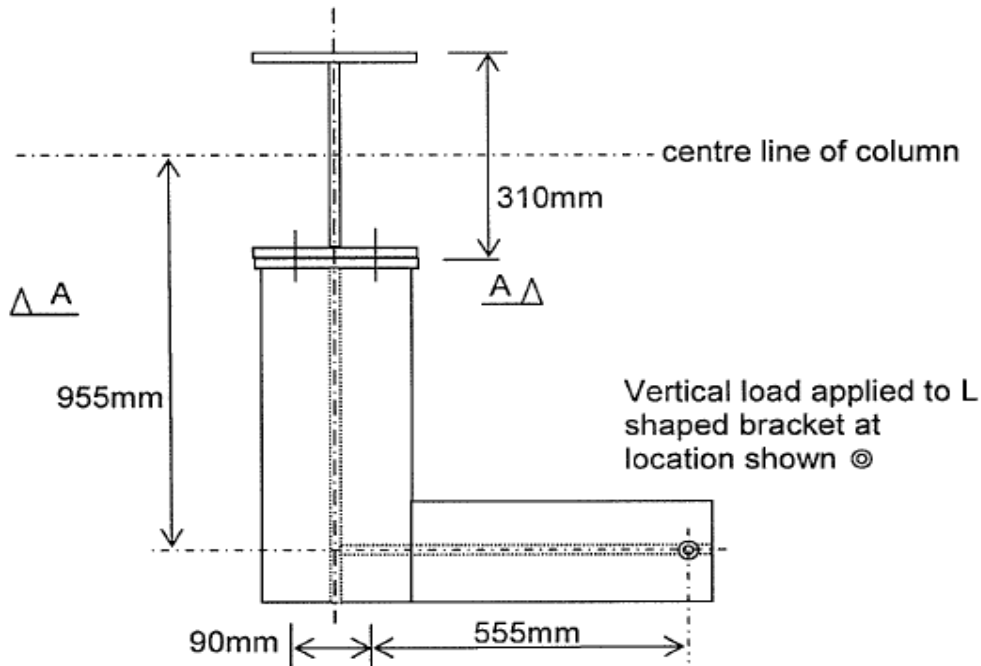


Figure Q4 (a)
 PLAN VIEW ON BRACKET

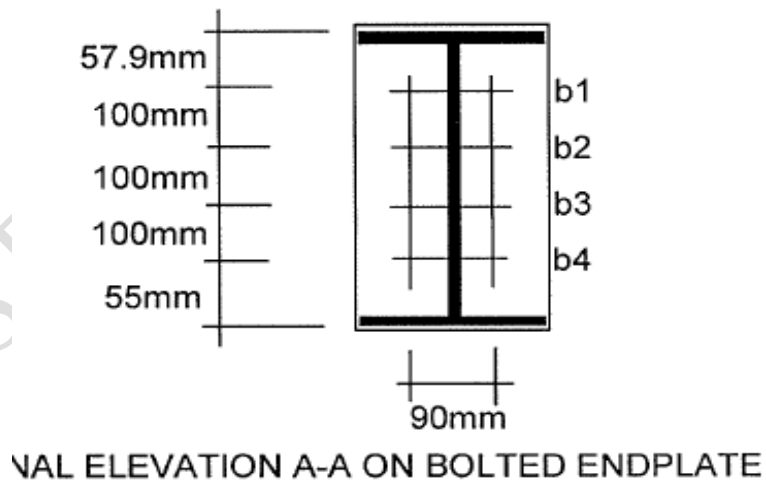


Figure Q4 (b)
 SECTIONAL ELEVATION A-A ON BOLTED ENDPLATE
 SHOWING SETTING OUT OF BOLTS

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Q5

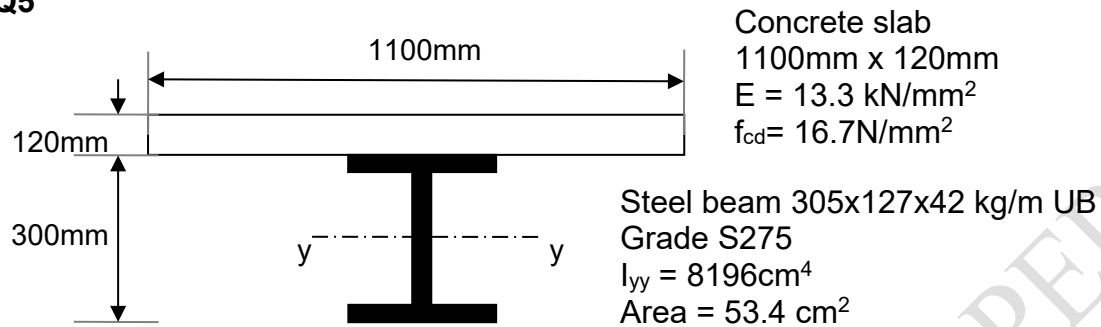


Figure Q5

Figure Q5 shows the section of a composite steel/concrete beam. The E value of the steel is 205 kN/mm² and the E value of the concrete is 13.3 kN/mm². The beam is simple supported over a span of 5.5m and carries the following factored uniformly distributed loads:

During construction (steel section alone carries loads)

10kN/m Dead Load + 15kN/m Imposed Load

In service (Loads are carried by the composite action)

15kN/m Dead Load + 18kN/m Imposed Load

- (i) Find the maximum working stress and maximum deflection of the beam during construction.

(5 marks)

- (ii) Transform the composite section to an equivalent steel beam. Find the position of the neutral axis, the value of the moment of inertia, $I_{y,comp}$, and the values of elastic section modulus, $W_{el,y,comp}$, for the transformed beam.

(12 marks)

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Q5 continued

(iii) For the in-service condition, find the maximum stress in the steel, the maximum stress in the concrete and the maximum deflection of the composite beam

(5 marks)

(iv) Check whether the stresses in steel and concrete are within the allowable limits.

(3 marks)

Total 25 marks

DATA

The central deflection of a simply supported beam carrying a uniformly distributed load

w per unit length is given by:
$$\delta = \frac{5wL^4}{384EI}$$

END OF QUESTIONS

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Extracts from Eurocode 3: Design of steel structures

6.3 Buckling resistance of members

6.3.1 Uniform members in compression

6.3.1.1 Buckling resistance

(1) A compression member shall be verified against buckling as follows:

$$\frac{N_{Ed}}{N_{b,Rd}} \leq 1,0 \quad (6.46)$$

where

N_{Ed} is the design value of the compression force
 $N_{b,Rd}$ is the design buckling resistance of the compression member.

(3) The design buckling resistance of a compression member should be taken as:

$$N_{b,Rd} = \frac{\chi A f_y}{\gamma_{M1}} \quad \text{for Class 1, 2 and 3 cross-sections} \quad (6.47)$$

$$N_{b,Rd} = \frac{\chi A_{eff} f_y}{\gamma_{M1}} \quad \text{for Class 4 cross-sections} \quad (6.48)$$

where χ is the reduction factor for the relevant buckling mode.

NOTE For determining the buckling resistance of members with tapered sections along the member or for non-uniform distribution of the compression force second-order analysis according to 5.3.4(2) may be performed. For out-of-plane buckling see also 6.3.4.

(4) In determining A and A_{eff} holes for fasteners at the column ends need not to be taken into account.

6.3.1.2 Buckling curves

(1) For axial compression in members the value of χ for the appropriate non-dimensional slenderness $\bar{\lambda}$ should be determined from the relevant buckling curve according to:

$$\chi = \frac{1}{\phi + \sqrt{\phi^2 - \bar{\lambda}^2}} \quad \text{but } \chi \leq 1,0 \quad (6.49)$$

where $\phi = 0,5 [1 + \alpha (\bar{\lambda} - 0,2) + \bar{\lambda}^2]$

$$\bar{\lambda} = \sqrt{\frac{A f_y}{N_{cr}}} \quad \text{for Class 1, 2 and 3 cross-sections}$$

$$\bar{\lambda} = \sqrt{\frac{A_{eff} f_y}{N_{cr}}} \quad \text{for Class 4 cross-sections}$$

α is an imperfection factor

N_{cr} is the elastic critical force for the relevant buckling mode based on the gross cross sectional properties.

(2) The imperfection factor α corresponding to the appropriate buckling curve should be obtained from Table 6.1 and Table 6.2.

Table 6.1 — Imperfection factors for buckling curves

| Buckling curve | a_0 | a | b | c | d |
|------------------------------|-------|------|------|------|------|
| Imperfection factor α | 0,13 | 0,21 | 0,34 | 0,49 | 0,76 |

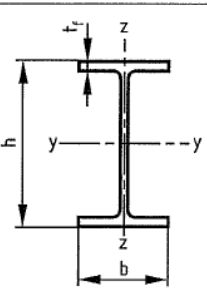
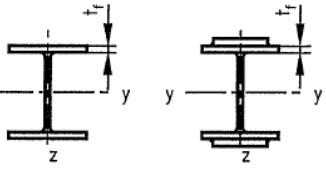

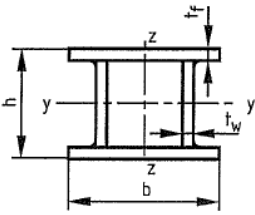
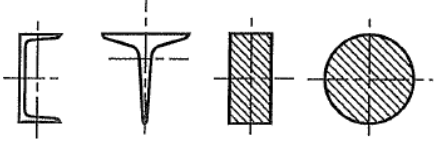
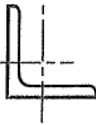
(3) Values of the reduction factor χ for the appropriate non-dimensional slenderness $\bar{\lambda}$ may be obtained from Figure 6.4.

(4) For slenderness $\bar{\lambda} \leq 0,2$ or for $\frac{N_{Ed}}{N_{cr}} \leq 0,04$ the buckling effects may be ignored and only cross-sectional checks apply.

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Table 6.2 – Selection of buckling curve for a cross-section

| Cross section | Limits | Buckling about axis | Buckling curve | | |
|--|---|---------------------|----------------------------------|--------|----------------|
| | | | S 235 S 275 S 355 S 420 | S 460 | |
| Rolled sections  | $h/b > 1,2$ | y - y z - z | $t_f \leq 40$ mm | a b | a_0 a_0 |
| | | | $40 \text{ mm} < t_f \leq 100$ | b c | a a |
| | $h/b \leq 1,2$ | y - y z - z | $t_f \leq 100$ mm | b c | a a |
| | | | $t_f > 100$ mm | d d | c c |
| Welded I sections  | $t_f \leq 40$ mm | y - y z - z | b c | b c | |
| | $t_f > 40$ mm | y - y z - z | c d | c d | |
| Hollow sections  | hot finished | any | a | a_0 | |
| | cold formed | any | c | c | |
| Welded box sections  | generally (except as below) | any | b | b | |
| | thick welds: $a > 0,5t_f$ $b/t_f < 30$ $h/t_w < 30$ | any | c | c | |
| U, T and solid sections  | | any | c | c | |
| L sections  | | any | b | b | |

**END OF SUPPLEMENTARY INFORMATION
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